FTR - Flight Test Report Dieser Prüfbericht darf ohne schriftliche Zustimmung der EAPR nicht, auch nicht a

Manufacturer		Type testing No.	
	Nova Vertriebsges.m.b.H Awweg 14 A-6123 Terfens	serial number	300050
Model	Triton 2 S	Landon	Brauneck
		Location	Schruns



Rev. 2.1 - 10.05.2013 EAPR GmbH - Marktstr. 11 D-87730 Bad Grönenbach - Germany

Date of testing	18.06.2014	Minimum take off 80 kg	weight	Maximum take off weight 100 kg			
Testpilot		Sepp Bauer		Hannes Tschofen	-		
Harness		EAPR- Testequipment		Academy Test Equipment			
Pilot's take off weigh	t	80 kg		100 kg	TAY AL		

Classification C



Test-criteria			Minimum take off weight			n Maximum take off weight	
1. Inflation / take-off - 4.1.1							
Rising behavior		Smooth, easy and constant rising		Α	Smooth, easy and constant rising		А
Special take off technique required		No		Α	No		Α
2. Landing - 4.1.2		•					
Special landing technique required		No		Α	No		Α
3. Speeds in straight flight - 4.1.3		•					
Trim speed more than 30km/h		Yes		Α	Yes		Α
Speed range using the controls larger than 10km	/h	Yes				Α	
Minimum speed		25 km/h to 30 km/	h	В	25 km/h to 30	km/h	В
4. Control movement - 4.1.4							
Max. weight in flight up to 80kg				-			-
Max. weight in flight 80 to 100kg		Increasing	45cm - 60cm	С	Increasing	45cm - 60cm	С
Max. weight in flight greater than 100kg				-			-
5. Pitch stability exiting accelerated flight - 4.	1.5						
Dive forward angle on exit		Dive forward less than 30° A			Dive forward I	less than 30°	Α
Collapse occurs		No		Α	No		Α
6. Pitch stability operating controls during acc	elerated 1	flight - 4.1.6					
Collapse occurs		No		Α	No		Α
7. Roll stability and damping - 4.1.7							
Oscillations		Reducing		Α	Reducing		Α
8. Stability in gentle spirals - 4.1.8							
Tendency to return to straight flight		Spontaneous exit	А	Spontaneous	exit	Α	
9. Behaviour in a steeply banked turn - 4.1.9							
Sink rate after two turns		More than 14m/s	В	More than 14m/s			
10. Symmetric front collapse - 4.1.10		•					
Entry	I	Rocking back less	than 45°	Α	Rocking back	less than 45°	Α
Recovery	trim speed	Spontaneous in less than 3 sec		Α	Spontaneous in 3 to 5 sec		В
Dive forward angle on exit	. <u>E</u>	0° - 30°	Geeping course	Α	30° - 60°	Entering a turn of less than 90°	В
Cascade occurs	=	No	1 0	A	No		A
Entry	р	Rocking back less	Rocking back less than 45°		A Rocking back greater than 45°		С
Recovery	lerate	Spontaneous in 3 to 5 sec 0° - 30° Entering a turn of less that		В	Spontaneous in 3 to 5 sec		В
Dive forward angle on exit	oce	0° - 30° E	Intering a turn of less than 90°	Α	30° - 60°	Entering a turn of less than 90°	В
Cascade occurs	ซี	No		Α	No		Α
11. Exiting deep stall (parachutal stall) - 4.1.1							

Flight Test Report -Musterprüfnummer: Seite 1 von 3

Deep stall achieved		Yes				Yes			
Recovery					Α	Spontaneous in 3 to 5 sec			С
Dive forward angle on exit		Spontaneous in less than 3 sec				Spontaneous in 3 to 5 sec			В
Change of course		0° - 30° Changing course less than 45°			A	Changing course less than 45°			A
Cascade occurs		No			A	No			A
12. High angle of attack recovery - 4.1.12					1				
Recovery		Spontaneous in le	ess than 3 sec		Α	Spontaneous in	3 to 5 sec		С
Cascade occurs		No			Α	No			Α
13. Recovery from a developed full stall - 4.1.1	3								
Dive forward angle on exit Collapse		0° - 30° No collapse			A	30° - 60° No collapse			B A
Cascade occurs (other than collapse)		No			A	No			A
Rocking backward		Less than 45°			Α	Less than 45°			Α
Line tension 14. Asymmetric collapse (trim speed) - 4.1.14		Most lines tight			Α	Most lines tight			A
	I			00 450				450 450	
Change of course until re-inflation	bse	< 90°	Dive or roll angle	0° - 15°	Α	< 90°	Dive or roll angle	15° - 45°	A
Re-inflation behavior	trim speed, max 50% collapse	Spontaneous re-in	Spontaneous re-inflation			Spontaneous re-		Α	
Total change of course	ds u	Less than 360°			Α	Less than 360°	Α		
Collapse on the opposite side occurs Twist occurs	tri Tax 5	No No			A	No No	A A		
Cascade occurs	=	No No			A	No No			A
Change of course until re-inflation	m	90° - 180°	Dive or roll angle	15° - 45°	В	90° - 180°	Dive or roll angle	45° - 60°	С
	trim speed, max 75% collapse		-flatiar				inflation:		
Re-inflation behavior	trim speed c 75% colla	Spontaneous re-in	ntiation		Α	Spontaneous re-	-inflation		Α
Total change of course Collapse on the opposite side occurs	rim s 75%	Less than 360°			A	Less than 360° No			A A
Twist occurs	max	No			A	No			A
Cascade occurs		No			Α	No			Α
Change of course until re-inflation	əsc	< 90°	Dive or roll angle	15° - 45°	Α	90° - 180°	Dive or roll angle	15° - 45°	В
Re-inflation behavior	accelerated, max 50% collapse	Spontaneous re-in	nflation		Α	Spontaneous re-inflation			Α
Total change of course	seler 0% o	Less than 360°			Α	Less than 360°			Α
Collapse on the opposite side occurs	acc lax 5	No			A	No No			A
Twist occurs Cascade occurs	Ε	No No			A	No			A
Change of course until re-inflation	Se	90° - 180°	Dive or roll angle	45° - 60°	С	90° - 180°	Dive or roll angle	45° - 60°	С
Re-inflation behavior	ated,	Spontaneous re-inflation			Α	Spontaneous re-	-inflation		Α
Total change of course	accelerated, x 75% collap	Spontaneous re-inflation Less than 360° No No		Α	Less than 360°			Α	
Collapse on the opposite side occurs Twist occurs	acc ax 7	No No			A A	No No			A
Cascade occurs	E	No No	No No			No No			A A
15. Directional control with a maintained asym	metric col	llapse - 4.1.15			Α				
Able to keep course straight		Yes			Α	Yes			Α
180° turn away from the collapsed side possible in 10 sec		Yes			Α	Yes			Α
Amount of control range between turn and stall or spin		25% to 50% of the symmetric control travel			С	25% to 50% of th	С		
16. Trim speed spin tendency - 4.1.16									
Spin occurs		No			Α	No			Α
17. Low speed spin tendency - 4.1.17 Spin occurs		l No			A No				٨
18. Recovery from a developed spin - 4.1.18		140			А	140			A
Spin rotation angle after release		Stops spinning in	less than 000		А	Stops spinning i	n 90° to 180°		С
Cascade occurs		No	1000 111111 30			No	11 00 10 100		
19. B-line-stall - 4.1.19		140			Α	140			A
Change of course before release		Changing course	less than 45°		Α	Changing course	e less than 45°		Α
Behaviour before release		Remains stable with straight span			Α	Remains stable without straight span			С
Recovery		Spontaneous in less than 3 sec			Α	Spontaneous in 3 to 5 sec			В
Dive forward angle on exit		0° - 30°			Α	30° - 60°	Α		
Cascade occurs		No			Α	No			Α
20. Big ears - 4.1.20		I .				l .			
Entry procedure		Standard technique			Α	Special device re	Α		
Behaviour during big ears		Stable flight			Α	Stable flight	Α		
Recovery		Recovery through pilot action in less than a further 3 sec			В	Recovery through pilot action in less than a further 3 sec			
Dive forward angle on exit		0° - 30°			Α	0° bis 30°			Α
21. Big Ears in accelerated flight - 4.1.21						1			
Entry procedure		Standard technique			Α	Special device required			
Behaviour during big ears		Stable flight			Α	Stable flight			Α
Recovery		Spontaneous in 3 to 5 sec			Α	Recovery through pilot action in less than a further 3 sec			В
Dive forward angle on exit		0° - 30°			Α	0° bis 30°			Α
Behaviour immediately after releasing the accelarator while maintaining big ears		Stable flight			Α	Stable flight			Α
				<u> </u>	<u> </u>				
22. Behaviour exiting a steep spiral - 4.1.22									

Flight Test Report - Musterprüfnummer: Seite 2 von 3

Tendency to return to straight flight	Spontaneous exit	Α	Spontaneous exit	Α					
Turn angle to recover normal flight	Less than 720°, spontaneous recovery	Α	A 720° to 1080°, spontaneous recovery						
23. Alternative means of directional control - 4.1.23									
180° turn achievable in 20 sec	Yes	Α	Yes	Α					
Stall or spin occurs	No	Α	No	Α					
24. Any other flight procedure and/or configuration desc	ribed in the user's manual - 4.1.24								
Procedure works as descibed		NA		NA					
Procedure suitable for novice pilots		NA		NA					
Cascade occurs		NA		NA					
25. Remarks of testpilot:									
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Flight Test Report - Musterprüfnummer: Seite 3 von 3